## OXYGEN SYSTEM MODIFICATION - ADJUSTMENT/TEST

### 1. General

- A. This procedure has the following tasks:
  - (1) Middle Pressure Leak Test Operational Test

NOTE: Test and check values for the following tests must be recorded in the tables provided.

- (2) Oxygen Pressure Reducer Test Operational Test
- (3) Simulated Automatic Actuation Test Operational Test

#### Middle Pressure Leak Test - Operational Test

- A. General
  - (1) This procedure ensures that the oxygen system res not have any pressure leaks.
- B. References
  - (1) AMM XX-XX-XX, Oxygen
  - (2) AMM XX-XX-XX, Air/Ground Relays
  - (3) AMM XX-XX-XX, Oxygen
  - (4) AMM XX-XX-XX, Passenger O: vste
  - (5) AMM XX-XX-XX, Flow Control U1.
- C. Access
  - (1) Location Zones
    - (a) XXX Forward Cargonia Compartment
    - (b) XXX For rgo tail. mpartment
    - (c)  $XXX \cup_{k}$  's. 'e
- D. Test Procedure
  - (1) Ens' the oxyge em is in serviceable condition (AMM XX-XX-XX).
  - (2) Ei aircraft i, ground mode (AMM XX-XX-XX).
  - (3) Re va fety precautions and general instructions before performing maintenant MM (3).
  - Open s panels to the oxygen cylinders.
    - MORE THAN 25 POUND-INCHES. THIS CAN CAUSE DAMAGE TO THE SHUT-OFF VALVE.
  - ne task "Close the Shutoff Valve on Passenger Oxygen Cylinders" (AMM XX-XX-XX, J. 201).
    - WARNING: LOOSEN SYSTEM COMPONENTS SLOWLY AND CAREFULLY. THE REMAINING OXYGEN CAN RELEASE WITH A LARGE FORCE AND CAUSE INJURY TO PERSONS AND DAMAGE TO THE AIRPLANE AND EQUIPMENT.
  - (6) Disconnect the supply manifold from the pressure reducer outlet at point A on each passenger oxygen cylinder (AMM XX-XX-XX, Fig. 501).

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- (7) Put a cap on each pressure reducer outlet to prevent contamination of the system.
- (8) Connect the auxiliary pressure source and a 0-2000 psi pressure gage to one supply manifold at point A.
- (9) Put a cap on the other open supply manifolds at point A.

WARNING: OPEN THE SHUT-OFF VALVE ON THE AUXILIARY PRESSURE SOURCE SLOWLY. HIGH TEMPERATURES CAN OCCUR, WHICH CAN START AN IGNITION WITH THE OXYGEN. THIS CAN CAUSE INJURY TO PERSONS AND DAMAGE TO THE AIRPLANE AND EQUIPMENT.

- (10) Open the shut-off valve on the auxiliary pressure source slow u star pressurize the system to a value of 1750 to 1950 psig.
- (11) Let the system cool down for 1 minute when the pressure re 1000 1500 psig.

CAUTION: DO NOT TIGHTEN THE SHUT-OFF VALVE ON THAT IT TEST CYLINDER MORE THAN 25 POUND-INC 'ES. THIS CAN C, SHUT-OFF VALVE.

- (12) After the system pressure becomes stable lose u. off on the auxiliary pressure source.
- (13) Make sure no leaks show over a p or nutes o. J-2000 psi pressure gauge.
- (14) If the gauge shows a pressure conformal following leak check:

WARNING: OPEN 7 7 SHU /ALVE ON THE AUXILIARY PRESSURE SOL' OWLY. TEMPERATURES CAN OCCUR, WHICH CAN STALL TION THE OXYGEN. THIS CAN CAUSE INJURY TO PL OLD DAN AGE TO THE AIRPLANE AND EQUIPMENT.

- (a) Start to the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly, and pressurize the system value of the auxiliary pressure source slowly.
- (b) Permit the s, to co wn for 1 minute when the pressure reaches 500, 1000 and psig.
- (c) ak of all ections for leaks with the leak detection compound.
- (d) up 'detection compound with a clean cotton cloth immediately after the
- (e) there are no leaks and test until the system is satisfactory.
  - VNG: DO NOT TIGHTEN THE SHUT-OFF VALVE ON THE PORTABLE TEST CYLINDER MORE THAN 25 POUND-INCHES. THIS CAN CAUSE DAMAGE TO THE SHUT-OFF VALVE.

Close the shut-off valve on the auxiliary pressure source.

- Return the system to normal:
  - WARNING: LOOSEN THE CONNECTION ON THE OXYGEN SYSTEM CAREFULLY. THE REMAINING OXYGEN CAN RELEASE WITH A LARGE FORCE AND CAUSE THE TEMPERATURE TO INCREASE. THIS CAN START AN IGNITION WITH THE OXYGEN AND CAUSE INJURY TO PERSONS AND DAMAGE TO THE AIRPLANE AND EQUIPMENT.

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- (h) Disconnect and remove the auxiliary pressure source, the pressure gauge, 0-2000 psi, and all caps from the oxygen system.
- (i) Reconnect the supply manifolds at point A (AMM XX-XX-XX, Fig. 501) on each oxygen cylinder.
- (15) Do the task "Open the Shutoff Valve on Passenger Oxygen Cylinders" (AMM XX-XX-XX).
- (16) Do AMM Task XX-XX-XX-702-079, Gaseous Oxygen System Leak Testing.
- (17) Close the access panels to the passenger oxygen cylinders.
- 3. Oxygen Pressure Reducer Test Operational Test
  - A. References
    - (1) AMM XX-XX-XX, Passenger Oxygen System
  - B. Test Procedure
    - (1) Do AMM task XX-XX-705-049-001, Oxygen Pressure Reduct ch passenger oxygen cylinder.
    - (2) Record results.
- 4. Simulated Automatic Actuation Test Operational Test
  - A. General
    - (1) This procedure ensures proper ment le oxygen system.

NOTE: Record result

- B. References
  - (1) AMM XX-XX-XX, 'ntegral \is, \ivsten.
  - (2) AMM XX-XX-X' rour, ala,
  - (3) AMM XX-XX gen stem
  - (4) AMM XX-XX-X Pens Oxygen Supply Components
  - (5) SSM XX-XX-XX, S Scric acs Manual
  - (6) WDM XX-XX, WILL agram Manual
  - (7) X' "ring Dia.
- C. Access
  - (1) Local
    - (a) Ad Cargo Container Compartment STA 720 to 970
    - /h) X per Half of Fuselage
  - tion
  - e that the electrical power is installed and serviceable (AMM XX-XX-XX).
  - (" ure that the oxygen system is installed and serviceable (AMM XX-XX-XX).
    - \_nsure that the airplane is in ground mode (AMM XX-XX-XX).
  - (4) Ensure that the Integrated Display System (IDS) is installed and serviceable (AMM XX-XX-XX).
  - (5) Ensure that the passenger address and entertainment systems are installed and serviceable (AMM XX-XX-XX).
  - (6) Read and obey the safety precautions and general instructions before performing maintenance (AMM XX-XX-XX).
  - (7) Supply electrical power (AMM XX-XX-XX, page 201).

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# Aircraft Maintenance Manual

(8) Remove the DO-NOT-CLOSE tag and close the following circuit breaker(s):

Panel	Circuit Breaker Title	Ident	Location
P7	OXYGEN RESET POWER		D3
P7	OXYGEN VALVE & IND		D4

- (9) Do the task Obtain Airplane Information Report (AMM XX-XX-XX, par > 201) to apply power to the EICAS (if necessary).
- (10) Do the task Access Bite Test (AMM XX-XX-XX, page 501) to turnecessary).
- (11) Open the panels to the oxygen cylinders.

CAUTION: DO NOT TIGHTEN THE SHUT-OFF VALVE ON 11 TO IEST CYLINDER MORE THAN 25 POUND-INCHES. THIS CAN MAGE TO THE SHUT-OFF VALVE.

(12) Close the shut-off valve on each oxygen cylinder.

NOTE: The shut-off valve can ' and clo. and, which is equivalent to 25 pound-inches.

WARNING: USF Y C SE, AN COMPONENTS IN THE OXYGEN SYSTEM. IF Y ON IN CLEAN COMPONENTS, THIS CAN CAUSE A SION WHEN NEAR PRESSURIZED OXYGEN. THIS CAN CAU MA EQUIPMENT OR INJURIES TO PERSONS.

- (13) If the stests will ducted using an auxiliary pressure source (as an alternative to texts. Let ye me cylinders), do the steps that follow:
  - (a) at point B from the coupling assembly of the oxygen cylinder to be set with a sk XX-XX-XX-715-128-001, Figure 501).

VING: USE ONLY OXYGEN CLEAN COMPONENTS IN THE OXYGEN SYSTEM. IF YOU DO NOT USE OXYGEN CLEAN COMPONENTS, A FIRE OR AN EXPLOSION CAN OCCURE. THIS CAN CAUSE DAMAGE TO EQUIPMENT OR INJURIES TO PERSONS.

Connect the auxiliary pressure source to the coupling assembly at point B (AMM Task XX-XX-715-128-001, Figure 501).

(c) Connect the auxiliary pressure source to a coupling assembly at the therapeutic oxygen bottles in the ceiling (AMM Task XX-XX-715-128-001, Figure 501, Point B).

NOTE: Oxygen clean fittings come from a sealed package labeled for oxygen system installation. Make sure that you use only oxygen clean fittings. Some fittings used in the oxygen system are the same as fittings used in other systems that are not oxygen clean. If it is necessary to clean parts, use the

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applicable oxygen procedures to clean the parts. This also applies to tube caps and plugs, which must be as clean as the installation connections.

WARNING: USE ONLY OXYGEN CLEAN COMPONENTS IN THE OXYGEN SYSTEM. IF YOU DO NOT USE OXYGEN CLEAN COMPONENTS, A FIRE OR AN EXPLOSION CAN OCCUR. THIS CAN CAUSE DAMAGE TO EQUIPMENT OR INJURIES TO PERSONS.

(d) Remove the plug at point C (AMM XX-XX-715-128-001, Figure 501) and connect a 0-2000 psig pressure gauge.

NOTE: Oxygen clean fittings come from 'led ' NOTE: age labeled for oxygen system installation. Make sure that y an fittings. oxyge Some fittings used in the oxygen system an me 📶 igs used in other systems that are not oxygen clean. If it o clean parts, . This also applies use the applicable oxygen procedures to clean to to tube caps and plugs, wh st be as clean as a installation connections.

- (e) Connect a 0-150 psig pressure 1 hos st port (pressure).
- (14) Restrain the oxygen box doors so ' ...e can un. Jut the oxygen masks do not drop.

NOTE: Use a tool, tar or oth able means to loosely restrain the doors to prevent the monday drop.

#### E. Test Procedure

(1) Make sure the hox e handle by tape or other applicable means.

NOTE: If the mask will then have to be installed again.

- (2) U ne to reic ne oxygen mask doors.
- (3) Op e on the auxiliary pressure source slowly and start to pressurize the system 'alu' 30 psig.
- (4) Conn able vacuum source to the M101 electropneumatic unit.

  Slowly ase the pressure on the variable vacuum source until the flow control unit comes

'k th. , lese things occur:

- The indicator on the flow control unit moves to ON position at a vacuum pressure of 18.11 to 17.30 inches Hg absolute (equivalent to 13.250 to 14.400 foot altitude). The 0-150 psig pressure gauge indicates an initial pressure surge (within 20 seconds) between 35 and 110 psig.
- (c) All the oxygen box doors unlatch.
- (d) The PASS OXYGEN ON advisory message displays on EICAS.
- (e) The NO SMOKING and FASTEN SEAT BELTS signs come on.
- (7) Slowly decrease the pressure on the variable vacuum source to 9.72 Hg absolute (equivalent to 28.000 foot altitude).
- (8) Make sure the pressure in the system is 25.10 to 35.17 psia.
- (9) Slowly decrease the pressure on the variable vacuum source to 5.54 inches Hg absolute (equivalent of 40.000-foot altitude).

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